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# USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

VOLUME 56 HC-13ON IN-FLIGHT CREW NOISE

**NOVEMBER 1975** 

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AEROSPACE MEDICAL RÉSEARCH LABORATORY
AEROSPACE MEDICAL DIVISION
Air Force Systems Command
Wright-Patterson Air Force Base, Ohio 45433

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FOR THE COMMANDER

HENNINGE. VON GIERKE
Director

Biodynamics and Bionics Division

Aerospace Medical Research Laboratory

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SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

A-weighted sound levels, preferred speech interference level, perceived noise level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application", AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

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### **PREFACE**

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 72310418, Measurement of Noise and Vibration Environments of Air Force Operations. Col Justus F. Rose, Jr. conducted the field measurements and performed the data analysis; Capt Nick Farinacci prepared this report.

The authors acknowledge the efforts of Mr. John N. Cole who established the data analysis requirements and assisted in the preparation of this report, and Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton who assisted in the mechanics of data processing.

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#### INTRODUCTION

The HC-130N is a USAF aircraft used to search for, locate and recover personnel and/or retrieve material in global air and space operations; it also serves as an air refueling tanker for helicopters. This aircraft, which is manufactured by the Lockheed Aircraft Corporation, Lockheed-Georgia Company, is powered by four T56-A-15 turboprop engines rated at 4,910 eshp at 13,820 rpm maximum take-off power. Each engine drives a Hamilton Standard four-blade constant-speed, 4.1 m diameter propeller through a 0.074 gear reduction. The engines are manufactured by the General Motors Corporation, Allison Division.

This volume provides measured data defining the bioacoustic environments produced inside this aircraft. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the HC-130N aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and aerospace ground equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, aerospace ground equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Refer to Volume 1 (reference 1) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., in-flight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published, and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of the updated index as it is generated.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; Autovon 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

#### IN-FLIGHT NOISE

#### **MEASUREMENTS**

All noise measurements were made on-board a standard-configured HC-130N aircraft during typical speed, altitude, and flight maneuver conditions. These levels describe the standard HC-130N environments, but may not be representative of those levels encountered if the aircraft has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made at various flight crew and passenger locations. Table 1 lists the measurement location and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A.

The microphone position was at ear level external to headgear in a region 0.2-0.3 meters from the head when an individual was present. At unoccupied locations, measurements were made at ear level throughout a volume where the head would normally be located. In both cases, the microphone was randomly moved throughout a spherical volume approximately 0.3 meter in diameter and the resultant samples analyzed using a 4- or 8-second integration time to obtain a power-averaged level that effectively smooths out short-duration fluctuations and best describes the exposure.

Although the presence of a crew member or passenger at a measurement location affects the resultant sound field, the magnitude of such effects is generally small and not significant in determining exposure limits or voice communication capabilities. Consequently, no distinction is made in this report between occupied and unoccupied measurement locations.

#### RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced inside the HC-130N aircraft at the 14 specified locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data, C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These variety of measures are widely used to assess the effects of noise on personnel and their performance.

## TABLE 1 MEASUREMENT LOCATIONS AND TEST CONDITIONS

HC-130N, Eglin AFB, 12 Aug 1971 Serial # 69-5829

LOCATION	POSITION	HEIGHT ABOVE DECK
1	Between Pilot and Copilot	Seated Head Level
2	Navigator Station	Seated Head Level
3	Radio Operator Station	Seated Head Level
4	Sta. 280 — Left Side	Seated Head Level

## TABLE 1 (Continued)

## MEASUREMENT LOCATIONS AND TEST CONDITIONS

## HC-130N, Eglin AFB, 12 Aug 1971 Serial # 69-5829

LOCATION	POSITION	HEIGHT ABOVE DECK
5	Sta. 280 — Centerline Standing at Desk	1.5 Meters
6	Sta. 280 — Right Side 2nd Engineer Seat	Seated Head Level
7	Sta. 420 — Left Side Propeller Plane	Seated Head Level
8	Forward Edge of Catwalk	1.5 Meters
9	Sta. 517 — Left Side Forward Bunk	Sleeping Height
10	Sta. 580 — Left Side Aft Bunk	Sleeping Height
11	Sta. 680 — Left Side	Seated Head Level
12	Sta. 680 — Centerline	1.5 Meters
13	Sta. 680 — Right Side	Seated Head Level
14	Sta. 260-337 — Right Side Bunk	Sleeping Height
CONDITION	DESCRIPTION	
A	Taxiing Torque — 2000 inlbs. Engine RPM — 100%	
В	Takeoff Torque — 19,600 inlbs. Engine RPM — 100%	
C	Initial Acceleration, gear and flaps up	
D	Climb Torque — 17,000 inlbs. Engine RPM — 100% Turbine Ivlet Temperature (TIT) — 900°C Indicated Airspeed (IAS) — 180 KIAS Altitude — 2.0 → 18.0M PA	
E	Cruise Torque — 12,500 inlbs. Engine RPM — 100% Turbine Inlet Temperature (TIT) — 960°C Indicated Airspeed (IAS) — 220 KIAS Altitude — 18.0M PA	
F	Descent Torque — 3000 — 4000 inlbs. Engine RPM — 100% Turbine Inlet Temperature (TIT) — 550°C Altitude — 18.0M PA > Indicated Airspeed (IAS) — 250 KIAS	
G	Descent Torque — 4000 inlbs. Engine RPM — 100% Turbine Inlet Temperature (TIT) — 550°C Indicated Airspeed (IAS) — 180 KIAS Altitude — 4.0 M PA	

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84 85 78 80 81 84 93 85 84 82 81 81 81 81 81 81 81 81 81 81 81 81 81	FREG												
8	(74)												
10	25		85	7.8	80		84	83	85	94	82	8	
92 93 82 82 86 94 91 93 87 87 87 89 89 89 89 89 89 89 89 89 89 89 89 89	1.5	88	88	8	87		90	90	88	99	85	80	
91 89 84 81 81 96 91 92 67 81 101 104 99 110 101 101 101 101 101 101 101 101	0	26	93	82	82		46	91	93	87	68	89	
103 95 102 92 89 111 104 99 110 101 101 101 101 101 101 101 101	0	91	89	48	81		96	91	92	87	85	82	
94  88  93  87  86  101  94  91  100  91  91  91  91  91  91  91  9	TT M	03	95	0	95		4	0	66	-	101	0	
90 87 86 87 88 87 86 87 88 86 89 86 89 80 80 80 80 80 80 80 80 80 80 80 80 80	0.0	76	88	93	87		0	76	91	8	91	9	
94         88         91         95         96         97         96         87         96         97         96         97         98         89         91         97         98         96         98         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96         96<	0	06	87	86	87		87	87	88	€	83	10 10 10 10 10 10 10 10 10 10 10 10 10 1	
86 84 85 86 87 86 89 88 91 94 94 89 88 89 89 89 89 89 89 89 89 89 89 89	10.	76	88	91	95		95	95	9.0	0	107	66	
87         87         86         87         88         96         95         86         96         95         86         96         95         96         95         86         88         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86         86<	ra (	86	4 1	8	80		96	89	88	91	96	9.0	
88 87 86 87 86 86 86 88 84 90 86 88 84 90 86 88 84 80 89 80 80 80 80 80 80 80 80 80 80 80 80 80		28	87	87	98		88	8 7	88	96	95	68	
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86 88 89 89 80 88 89 84 86 86 86 86 86 86 86 86 86 86 86 86 86		60	06	80 G	06		80	90	91	92	86	88	
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81 82 78 84 78 82 83 75 81 8 79 80 76 80 82 72 73 73 77 71 79 78 72 76 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 77		99	28	4	98		3	86	88	81	96	87	
79     80     77     73     77     79     75     79     78     72     79     78     72     76     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     77     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     78     <	5.1	81	28	7.8	85		78	82	83	15	81	83	
74 75 72 77 71 78 76 77 77 77 74 77 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 74 77 77	5	62	0 1	92	80		92	1 6	85	73	4	80	
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70 70 70 70 71 72 68 72 71 73 75 75 77 70 70 69 69 68 67 58 67 73 75 75 75 70 69 69 65 69 67 58 67 73 75 75 75 60 69 65 69 65 71 69 67 69 69 66 75 77 60 69 69 66 75 77 60 60 65 65 65 65 65 65 65 65 65 65 65 65 65		73	73	7.1	73		9	7.0	7.	73	12	12	
70 69 70 70 71 68 71 70 73 76 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7		20	7.0	7.0	7.0		68	72	7.1	73	75	7.	
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0 69 65 71 69 67 69 69 66 76 78 6 0 66 63 69 67 64 68 67 63 74 75 6 0 65 63 67 65 63 65 61 69 71 6 0 64 63 67 64 63 65 64 62 67 69 6 10 105 101 104 101 110 112 106 103 111 108 10	•	99	65	7.1	69		68	69	99	25	11	99	
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				0	0	_	-			•	_	_	
		-	1	• 1	1	1	• •	, 1	•	• •	, (	) (	

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

2	MEASUKED SOUND PRESSURE LEVEL OCTAVE BAND	RESSUR	ב רבענו	(60)								) IDENTIFICATION: ) OMFGA 3.2
NOISE SOURCE/SUBJECT: HC-130N AIRCRAFT INFLIGHT NOISE LEVELS	ECT: T LEVELS		OPERATION:	 					! ! !			-) TEST 71-014-058 ) RUN 0: ) 02 JAN 75
FREQ (HZ)	1/A	1/8	1/0	170	1/5	LOCATION/CONDITION 1/G 2/A 2/D	N/COND 2/A	1110N 2/0	2/E	2/6	3/A	3/0
31.5	95	109	102	89	06	60	0.6	3	C	4	6	78
63	93	103	104	102	105	96	89	95	106	96	16	107
125	87	95	91	88	93	91	98	91	46	93	80	56
250	95	90	9.0	85	87	88	76	96	90	88	95	36
500	96	86	83	83	87	88	9.0	84	38	88	89	86
1000	92	79	75	92	81	82	77	9/	83	83	11	7.8
2000	69	74	72	7.1	14	75	69	72	92	75	69	73
000	49	7.0	69	68	69	7.0	99	7.1	69	7.1	29	71
8000	29	73	72	99	63	49	6.8	7.0	65	29	68	71
16000	58	65	9	25	59	9	61	61	62	63	62	99
OVERALL	100	-	107	4 0 2	4	o	60	6	,	Ġ	Ċ	

2 OCTAVE BAND	BAND	KE 330 K	MEASUREU SOUND PRESSURE LEVEL OCTAVE BAND	(08)								) IDENTIFICATION: ) ) OMEGA 3.2
NOISE SOURCE/SUBJECT:	JECT 8		OPERATIONS	1 2 N								-) TEST 71-014-058 ) RUN 02 ) 17 JAN 75
INFLIGHT NOISE LEVELS	LEVELS											) PAGE J2
FREQ (HZ)	3/E	3/6	0/1	4/E	4/F	0CAT 10 5/D	LOCATION/CONDITION 5/D 5/E 5/F	ITION 5/F	0/9	6/E	6/F	7/0
31.5	86	60	06	91	ec	87	78	98	93	06	91	60 60
63	108	26	116	120	109	101	106	93	115	117	101	110
125	95	93	103	105	102	102	66	98	101	102	105	103
250	93	88	91	96	<b>1</b> 6	95	98	76	35	101	<b>7</b>	26
500	26	89	88	93	76	89	95	76	88	95	76	06
1000	98	92	81	87	88	80	86	96	81	98	8.7	84
2000	75	47	6.2	83	83	79	83	81	80	81	81	86
4000	20	7.1	80	81	7.8	81	82	7.7	81	6.2	7.8	187
8000	29	71	81	82	73	82	83	71	81	79	4 ~	06
16000	29	99	73	52	71	74	15	6.8	92	72	4.2	85
OVERALL	109	100	116	120	7		- C	4 0 2	116	117	404	• • • •

2 OCTAN	DCTAVE BAND	RE SSUR	E LEVEI	(08)								) IDENTIFICATIONS ) OMEGA 3.2
NOISE SOURCE/SUBJECT	RCE/SUBJECT:	  -  -	OPERATIONS	10N			~ -	• • • •				) IEST 71-014-058 ) RUN 03
HC-130N AIRCRAFT INFLIGHT NOISE LEVELS	CRAFT ISE LEVELS											) 02 JAN 75
		-					^					) PAGE 33
	7/E	7/F	0/9	8/E	8/F	OCATIC 9/0	LOCATION/CONDITION 9/D 9/E 9/F	NITION 9/F	10/D	10/5	10/F	11/0
FREQ (HZ)												
31.5	68	9.0	80	87	87	92	91	92	93	91	6	92
63	111	108	111	108	102	116	107	106	110	106	109	101
125	104	<b>1</b> 6	66	95	46	96	104	93	96	105	95	93
250	95	93	96	95	92	93	95	91	93	<b>3</b> 6	46	92
200	36	93	68	91	92	88	8	68	89	91	93	91
1000	88	96	82	88	48	83	82	82	81	48	98	80
2000	87	81	82	40	79	91	42	78	78	8 0	8 0	76
0004	80	78	<b>†</b> 0	98	92	83	4	74	77	11	7.4	73
8000	06	77	87	86	75	98	82	73	4	11	72	74
16000	85	72	81	7.9	29	7.9	4.2	29	74	7.2	69	69
OVEDAL	110	401		0	70		00+	404	4	•	9	~ 0 +

2 OCTAVE	OCTAVE BAND		PRESSURE LEVEL	(60)								) IDENTIFICATIONS ) OMEGA 3.2
NOISE SOURCE/SUBJECT:	BJECT	-	OFERATION:	10N 8			-					-) TEST 71-014-058 ) RUN 04
HC-130N AIRCRAFT INFLIGHT NOISE LEVELS	AFT E LEVELS											) 02 JAN 75 ) PAGE J4
FREQ (HZ)	11/E	11/F	12/0	12/5	12/F	137D	LOCATION/CONDITION 13/D 13/E 13/F	DITION 13/F	14.70	14/E	14/F	
31.5	76	95	8.7	89	89	95	93	96	91	91	06	
63	104	96	103	93	91	112	105	100	110	101	101	
125	96	91	6	96	95	*6	96	93	102	107	100	
250	36	92	95	92	46	95	95	93	96	96	93	
500	95	93	95	93	95	95	93	76	88	91	36	
1000	9.4	85	81	85	87	80	86	98	7.8	48	85	
2000	79	90	77	79	81	16	82	90	7.8	81	4	
000	7.4	7.4	47	14	25	72	75	7.4	7.8	8.0	92	
8000	73	69	75	73	7.1	73	73	7.0	90	81	7.0	
16000	29	99	7.0	68	99	68	19	49	71	73	69	
OVERALL	105	101	104	101	100	112	106	104	1 4 4	40	4	

m												A 3.2
NOISE SOURCE/SUBJECT! HC-130N AIRCRAFT INFLIGHT NOISE LEVEL	ELS		OPERATION:	z O	 	1 1 1 1		1 1 1 1 1				) TEST 71-014-058 ) RUN 01 ) 02 JAN 75 ) PAGE H4
	1/A	1/8	1/0	170	1/E	LOCATIO 1/6	CATION/CONDI 1/G 2/A	OITION 2/0	2/E	2/6	3/A	, [
HAZARD/PROTECTION C-WEIGHTED OVERALL A-WEIGHTED OVERALL MAXIMUM PERMISSIBLE NO PROTECTION	Sou Sou	S 5 7	LEVEL (OA LEVEL (OA (T IN MIN	(OASLC IN (OASLA IN MINUTES)	N 08C) / N 08A) / FOR ON	7 7 W	F EAR F EAR Exposure	PER DAY	(AFR	161-35,	J UL Y	73)
	66	107	105	102	105	86	97	76	106	<b>6</b> 6		107
		285	339	100	240	240	202	t 0	170	240	0 <b>7</b> 2	285
HGU-ZAZP HELMET WIT	WITH H-1	54	6	q	•	9	3	G	ě			
-		571	571	960	571	807	407	96.0	900	200	\$ C	J C
HGU-ZA/P HELMET WI	7	_										
( ) <del>-</del>	629	807	807	960	807	960	807	96.0	571	96.0	0.00	5.0 S
ELMET	WITH CUS	TOM L	INER				)				,	
OASLA	80 %	80 5	400	00 00	96	80 c	88 6	85	80 6	986	986	86
V-51R EAR PLUGS	1	† •	0	6/9	200	<b>*</b>	0 \$ 7	619	0 \$ 2	5.59	539	839
	69	69	68	65	6.8	99	6.8	63	7.0	99	29	2.0
THE CHAPTER TO SERVE	960	•	960	960		960	096	960	960	096	096	096
	COMMON IC		- 08 0 8	77	80	75	7.6	74	82	75	75	282
_	096	960	966	960	096	960	096	096	619	096	. 0	679
COMMUNICATION PREFERRED SPEECH PSIL	INTER 78	FERENCE 80	SE LEVEL	(PS	IL IN	08)	62	7.2	80 80	280	5	62
IVED NOISE			CORRECTED		(PNLT IN	N PN08)	_					
	103 103	105	105	103	106	102	102	101	108	103	102	107

												OMEGA 3.2
NOISE SOURCE/SUBJECT	CT 1	Ŭ.	OPERATION:	I NO			~					) TEST 71-014-058 ) RUN 02
HC-130N AIRCRAFT	( :	<b>.</b>					~ ~					) 02 JAN 75
INFLICATION DISE LEVELS	EVELS						~ ~					) FAGE H2
	3/E	3/6	0/4	4/E	4 / F	LOCATION/CONDITION 5/0 5/E 5/F	N/CONE 5/E	SITION 5/F	0/9	6/E	6/F	7/0
HAZARD/PROTECTION C-WEIGHTED OVERALL SOU A-WEIGHTED OVERALL SOUMAYTHIM DEDMICSTOLF TO	ON OVERALL SOUND OVERALL SOUND		LEVEL (OA)	COASLC IN	1 08C) A 1 08A) A	<b>← ←</b>				,		į
NO PROTECTION	31046 11			1631			CAPUSURE	FEK DAY	A T X	161-35,	JUL *	(3)
DASLC	108	66	116	119	109	105	107	101	115	116	107	111
DASLA	93	80	9 C	86,	95	26	76	76	76	96	95	96
P HELMET	MITH H-1	202	0	<b>*</b>	כ	120	32	82	80 22	9	7.1	09
	87	82	95	96	9.0	89	9.0	89	6	φ υ	5	60
	285	629	120	09	170	202	170	240	120	7.1	143	120
HOU-ZAZIV HELMEI	MIIH H-1	54(A)	ć	L	į		1		1	П		
CASCA	0 5	0 0	16	, t	10 0	100	9 1	M (	06	93	86	88
HGU-ZA/P HELMET	WITH CUST	ON	INER	1/	687	2 2 3	339	277	170	101	339	240
	06	86	93	26	93	9.0	92	91	35	95	26	25
-	170	339	101	50	101	170	120	143	120	7.	101	120
V-51R EAR PLUGS	ļ		į				1		;	•		,
UASLA	5 2	29	7.8	8	74	7.0	73	7.1	11	79	73	7.4
H-157 IN-FLIGHT	COMMUN TC	ATTON	12 P D D T T T T T T T T T T T T T T T T T	>609	096	960	096	960	960	960	960	960
	9.4	76		95		8	40	ec.	σ	0	7	8.7
_	480	960	143	7.1	339	571	180	960	170	120	480	285
COMMUNICATION PREFERRED SPEECH PSIL		INTERFERENCE 85 82	SE LEVEL 83	PSIL 88	1 N 8 8	08)	28	87	83	9	87	87
ANNOYANCE PERCEIVED NOISE TOME CORRECTION	E LEVEL,	TONE	CORRECTED		(PNLT IN	90N9 7						
PNLT		103	116	120	113	111	112	109	115	117	113	r u

BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE. TIME LIMIT SET TO AVOID WHOLE 900Y EFFECTS (WHOLE BODY LIMITS EXTRAPOLATED AT -4 08 PER DOUBLE TIME).

TABLES MEASURES OF HUMAN	N NOISE	EXPOSURE	URE								IFICATIO A 3.2
E/SU	Ŭ.	OPERATION	I NO			^					-) IESI 71-014-058 ) RUN 03
HC-130N AIRCRAFT						• •					) 02 JAN 75
INFLIGHT NOISE LEVELS				;  -							
7/E	7/F	8/0	8/E	8/F	LOCATION/CONDITION 9/D 9/E 9/F	N/CON0 9/E	1	10/0	10/E	10/F	11/0
				080)	AT EAR						
( A-WEIGHTED OVERALL SOUND MAXIMUM PERMISSIBLE TIME NO PROTECTION		LEVEL (OA	(OASLA IN Minutes)	DBA) FOR ON	-	SURE	PER DAY	(AFR	161-35,	, JULY	73)
	107	110	108	103	116	108	106	109	108	108	102
( 0ASLA 97	93 101	9 <b>2</b>	9 6	92 120	95	93	90 170	91	9 9 9	93	91
ELMET WITH	54							1 3		į	
C DASLA+ 92	785	90	888	86	95	06	80 2	80 6	91	e0 :	20.00
ELMET WIT	-154(A)	•	) }	600	150	1 / 0	*	7	1 4 3	n *	<b>*</b>
LA*	85	87	92	82	91	98	83	86	87	85	82
240 T HGH=2478 HGH HET HITH CHS	404	285	707	619	143	339	571	339	285	707	629
	90	1 NEK 91	9.0	89	93	91	8.8	89	95	91	9D
T 101	170	143	170	202	101	143	240	202	120	143	240
( V-SIR EAR PLUGS 75	72	7.4	72	2.0	7.8	72	7.0	72	7.3	7.3	6
_	096		960	096	960	096	960	096	960	096	960
( H-157 IN-FLIGHT COMMUNIC ( OASLA* 88	A	5	48	8.0	91	#C	8.1	5	86	4	62
2	57.1	339	480	096	143	404	807	404	339	480	096
C COMMUNICATION  PREFERED SPEECH INTERFE	RFEREN		L (PSIL	Z	08)						
	86	94		8 5	9.4	83	83	83	85	86	83
( ANNOYANCE PERCEIVED NOISE LEVEL, TONE CORDECTION OF IN	-, TONE	CORRECTED		(PNLT IN	PN08)						
PNLT 117	111	114	112	107	116	112	108	111	113	109	105 1
# BASED ON CALCULATED SPL	t	SPECTRUM	UNDER	PROTECTIVE		DEVICE.					***************************************

HC-130N AIRCRAFT  INFLIGHT NOISE LEVELS  INFLIGHTED TO STATE TO	040 B40	CCO	13/F 13/F 103 93 101		14/E 161-35 108 95 71	14/F 104 104 104 101 101	T 71-056
LEVELS ( 11/E 11/F 12/D 12/E 1 11/E 11/F 12/D 12/E 1  ERALL SOUND LEVEL (OASLC IN ERALL SOUND LEVEL (OASLA IN SSIBLE TIME (T IN MINUTES) F 120 120 143 101 WITH H-154	2 0 0 0 1 W V	0N/COND 13/E 13/E 105 B 101 105	11110N 137F 103 103	(AFR 1110 92 120 91	14/E 161-35 108 95 71	14/F 104 101 101	A H I
11/E 11/F 12/D 12/E 1  ERALL SOUND LEVEL (OASLC IN ERALL SOUND LEVEL (OASLA IN SSIBLE TIME (T IN MINUTES) F  105 100 103 100 92 92 91 93 120 120 143 101 WITH H-154	04 0 0 1 0 N	13/E 13/E 13/E 105 105 101	137F 137F 103 103	(AFR 1110 92 120 91	14/E 161-35 108 95 71	14/F 104 93 101	I
11/E 11/F 12/D 12/E 1  ERALL SOUND LEVEL (OASLC IN  ERALL SOUND LEVEL (OASLA IN  SSIBLE TIME (T IN MINUTES) F  105 100 103 100  92 92 91 93  120 120 143 101  MITH H-154	0 0 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0 1	13/E 13/E 13/E 0SURE P 105 93	13/F 13/F 103 103	(AFR 110 92 120 91	161-35 108 95 71	14/F 104 93 101	
ERALL SOUND LEVEL (OASLC IN ERALL SOUND LEVEL (OASLA IN SSIBLE TIME (T IN MINUTES) F 105 100 103 100 92 92 91 93 120 120 143 101 WITH H-154	STO ON E	105 P	0 A 0 B 0 A 0 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	~	161-35 108 95 71	104 104 93 101	
A-WEIGHTED OVERALL SOUND LEVEL (DASLA IN MAXIMUM PERMISSIBLE TIME (T IN MINUTES) F PROTECTION  OASLC  OASLC  105  100  120  143  101  U-2A/P HELMET WITH H-154	AD ONE	105 105 101 101	0 A D D D D D D D D D D D D D D D D D D	~	161-35 108 95 71	104 104 101	
0ASLC 105 100 103 100 0ASLA 92 92 91 93 T 120 120 143 101 U-2A/P HELMET WITH H-154		105 101 101	103 101	110 92 120	108 95 71	104 93 101	
92 92 91 93 120 120 143 101 WITH H-154		101	101	92 120 91	71	101	
120 120 143 101 WITH H-154		101	101	120 91	7.1	101	
CT-U UITM		18	,	91		0	
86 85 85 85			2		0		
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ANNOYANCE PERCEIVED NOISE LEVEL, TONE CORRECTED (PNLT IN PNDB) TONE CORRECTION (C IN DR)	T IN PNOB	2					
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